

## Application of the Jaya algorithm to solve the optimal reliability allocation for reduction oxygen supply system of a spacecraft

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### Article Info

#### Article history:

Received Dec 23, 2020

Revised Sep 5, 2021

Accepted Sep 11, 2021

#### Keywords:

Jaya algorithm

Optimal reliability allocation

Reduction oxygen

Reliability network

Reliability optimization

Spacecraft

### ABSTRACT

In this paper the reliability of reduction oxygen supply system (ROSS) of a spacecraft which was calculated as a complex system using minimal cut method. The reliability of each component of system was calculated as well as the reliability importance of the system. The cost of each component of the system was possible approaches of the allocation values of reliability based the minimization of the overall cost in this system. The advantage of this algorithm can be used to allocate the optimization of reliability for simple or complex system. This optimization is achieved using the Jaya algorithm. The proposed technique is based on the notion that a conclusion reached on a particular problem should pass near the best results and avoid the worst outcomes. The original findings of this paper are: i) the system used in this paper is a spacecraft's reduced oxygen supply system with the logarithmic cost function; and ii) the results obtained were by using the Jaya algorithm to solve specific system reliability optimization problems.

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## 1. INTRODUCTION

One of the first researchers to take up the reduction oxygen supply system (ROSS) study of a spacecraft system for the first time by Aggarwal [1], and due to the importance of the topic in terms of engineering and science, light was re-shed by several researchers such as Hassan *et al.* [2]–[9] where the researchers discussed the reliability of this system and the study of mean time to failure (MTTF) using the engineering properties of many reliability limits. Also Abed *et al.* [10] discussed the issue of optimizing the reliability of this system has solved by using a genetic algorithm.

A ROSS of a spacecraft consists of a group of subsystems that are interconnected in a specific way that requires the designer to either maximize the reliability of the system while reducing the total cost or reduce the costs to the lowest possible level with appropriate reliability so that the system works in a suitable performance and in a specific period of time due to the current budget only. Many researchers have addressed the problem of optimizing the reliability of this in electrical or mechanical and electromagnetic systems and

in many fields [11]–[19]. The complexity or reliability appropriate for the system's continuous operation for a specific period of time and there are other factors that the specialist considers appropriate [20]–[24].

This paper studies the problem of optimal allocating reliability as a mathematical problem even though problems roots belong to network. This paper deals with the problem of optimization of the device ROSS [25]–[28]. The system consists of subsystems or components where the reliability requirements between the subsystems or components are determined based on the importance of reliability or the location of the component in the system. The model for a component's reliability allocation is based on the cost of increasing reliability that's own. The costliest elements will be (the cost expressing by weight, cost, size, or any other amount) increases in reliability. By using these algorithms, it is possible to assign reliability to some or all of the components of any system, whether that system is complex or not [29]–[33]. Parameters are possible to be change the of the proposed cost function according to business requirements or the actual need for the system to function well, which will have engineers examine allocation scenarios for reliability in all respects before making decisions. The logarithmic cost function is increasing and convex. Also, the results obtained after solving the optimization problem using the Jaya algorithm [34]–[68] contributed to improving the optimization of the system reliability with appropriate costs [33], [69]–[80].

## 2. RESEARCH METHOD

### 2.1. Optimization for ROSS of a spacecraft

Consider a ROSS of a spacecraft consisting of elements connected reliability [15], [81]–[84], the following notes are used:

- $0 \leq R_i \leq 1$  is the element reliability  $i$
- $C_i(Q_i)$  is the cost of the element  $i$
- $C(Q_1, \dots, Q_n) = \sum_{i=1}^n a_i C_i(Q_i)$  the system's total cost, where  $a_i > 0$
- $Q_i$ : is unreliability
- $Q_s$ : is the unreliability of the system
- $Q_G$ : is the unreliability goal of the system

The main goal of the problem is to allocate reliability to some or all components of the system to reduce costs to the least possible. The problem P will be formulated as the objective function (cost function) and the constraints as the optimization problem in nonlinear programming as (1):

$$P: \text{Find Minimize } C(Q_1, \dots, Q_n) = \sum_{i=1}^n a_i C_i(Q_i), a_i > 0 \tag{1}$$

$$\text{subject to } (1 - Q_s) \geq (1 - Q_G) \quad 0 \leq Q_i \leq 1, i = 1, 2, \dots, n$$

where  $R_i + Q_i = 1$ , assuming  $C_i(Q_i)$  the cost function be positive and differentiable [21]:  $\left[ \Rightarrow \frac{dC_i}{dQ_i} \geq 0 \right]$ . The cost convexity of Euclidean function is to minimize the cost function. This is the same as the derivative  $\frac{dC_i}{dQ_i}$  is monotonically increasing, i.e.,  $\frac{d^2C_i}{dQ_i^2} \geq 0$ . The purpose of the previous plan is to achieve an all-out framework cost base [4], subject to  $R_G$ , the lower system reliability limit.

### 2.2. Model of logarithmic function

Let  $0 < Q_i < 1$  and  $K_i$ , constant,  $i = 1, 2, \dots, n$ , the cost function has been proposed in a number of references. The most important of which are in the form formula give the following cost unreliability curve as (2).

$$c_i = K_i \ln \left( \frac{1}{1 - (1 - Q_i)} \right), K_i > 0 \tag{2}$$

Consequently, each  $C_i(Q_i)$  it is a monotony increasing and convex function in. The cost of the entire  $C(Q_1, \dots, Q_n) = \sum_{i=1}^n a_i C_i(Q_i)$  has similar characteristics [3].

### 2.3. Jaya algorithm

Jaya algorithm can solve optimization problems and it is one of the methods used to solving linear and nonlinear problems (with constraint or unconstraint), and this algorithm has been applied in many of the most difficult optimization problems [29], [74]. An important difference between Jaya and the other algorithm in optimization is that parameters of Jaya does not require tuning (a process required in other algorithms to avoid unwanted convergence). Another difference is Jaya's efficiency in always choosing the best solution. Next is the steps of Jaya algorithm and Figure 1 shows in flowchart form.

- Step-1 : Set population size (p-size), unreliability, mutation unreliability (p-mute), maximum generation (max-gen) and bounds of the variables.
- Step-2 : Initialize the unreliability of individual components  $Q_i$  [ $Q_i$  represents the component at i-th generation].
- Step-3 : Evaluate each component's cost function  $c_i(Q_i)$  consider the function of an objective as a cost function.
- Step-4 : Find the component's lowest unreliability  $Q_i$ .
- Step-5 : Go to Step-12 if the termination criterion is met, otherwise go to the next step.
- Step-6 : Component selection  $Q_i$  component iteration j  $Q_j$  iteration j, by tournament selection process of the earlier generation.
- Step-7 : Change the unreliability  $Q_i$  by operators of crossover, mutation and elitism.
- Step-8 : Assess the cost function value of each  $Q_i$ .
- Step-9 : Find each component's best reliability
- Step-10 : Compare and better store the lowest unreliability of each  $Q_i$  iteration component j and the unreliability of each component iteration j  $Q_j$ .
- Step-11 : Print the best unreliability of each component (Which is the solution to the problem of optimization).
- Step-12 : Calculate the Reliability of each component  $R_i$  by the equation  $R_i + Q_i = 1$ .
- Step-13 : The end.

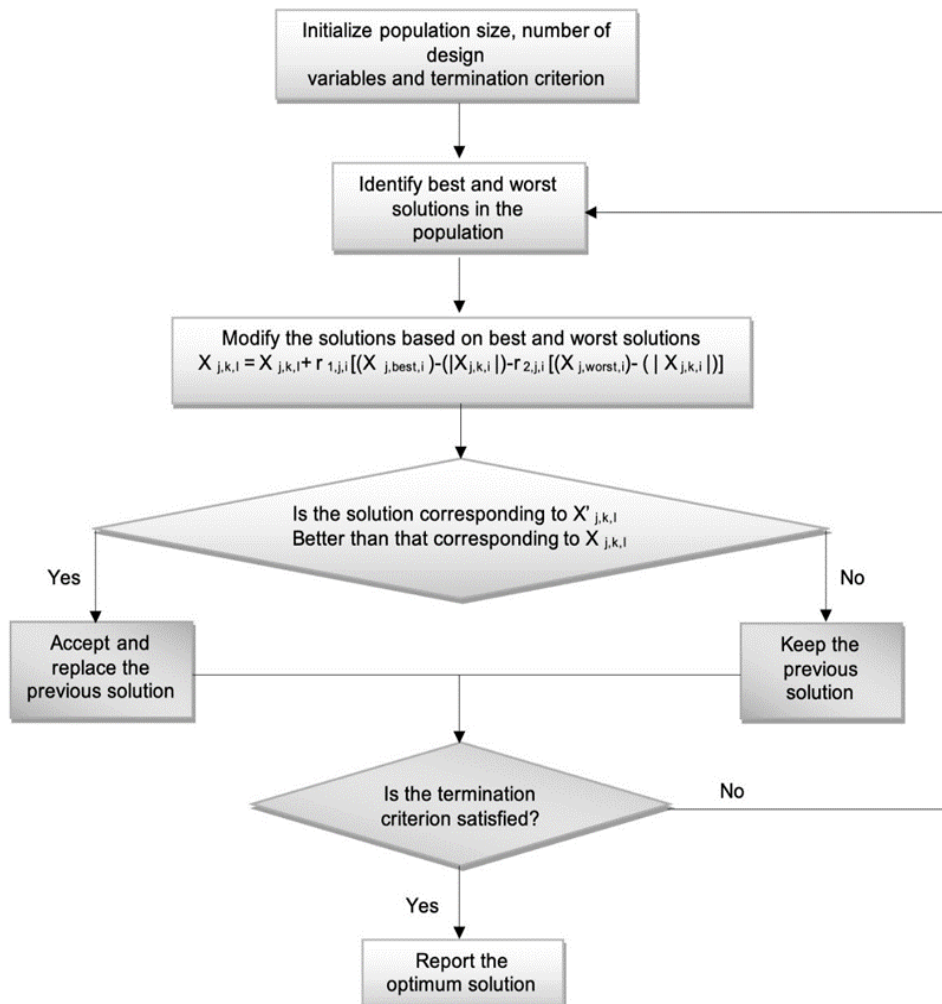


Figure 1. Flowchart of the Jaya algorithm

**3. RESULTS AND DISCUSSIONS**

Let us have a ROSS system consisting of 11 components, which is as shown in Figure 2 and after reducing the number of components, the number of components became 7 as shown in Figure 3, where (check valve, shut off valve, non-return automatic emergency valve) was connected in parallel in one component, so that the component number was (5) as well in the same way for the component number (6), has the same primary unreliability in all components that have at 0.2% at a specified time.

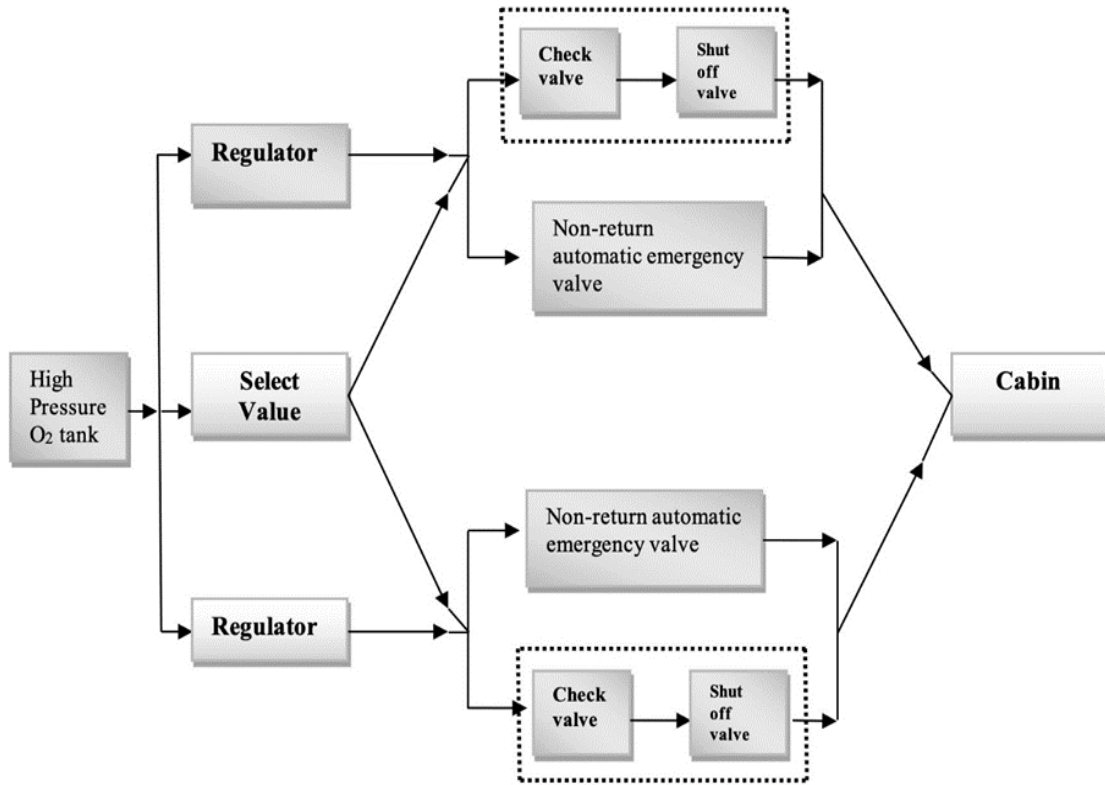


Figure 2. Oxygen supply system of a spacecraft

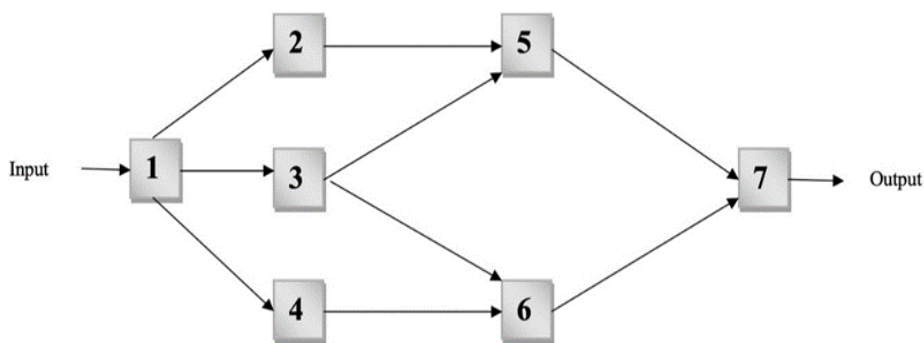


Figure 3. ROSS of a spacecraft after reducing

The system unreliability objective at a specified time is 0.1%. The polynomial unreliability of the given system was calculated using the minimal cut method as (3).

$$\begin{aligned}
(1 - Q_s) = & (1 - Q_1)(1 - Q_2)(1 - Q_5)(1 - Q_7) + (1 - Q_1)(1 - Q_3)(1 - Q_5)(1 - Q_7) \\
& + (1 - Q_1)(1 - Q_3)(1 - Q_6)(1 - Q_7) + (1 - Q_1)(1 - Q_4)(1 - Q_6)(1 - Q_7) \\
& - (1 - Q_1)(1 - Q_2)(1 - Q_3)(1 - Q_5)(1 - Q_7) \\
& - (1 - Q_1)(1 - Q_3)(1 - Q_4)(1 - Q_6)(1 - Q_7) \\
& - (1 - Q_1) - (1 - Q_1)(1 - Q_3)(1 - Q_5)(1 - Q_6)(1 - Q_7) \\
& - (1 - Q_1)(1 - Q_2)(1 - Q_4)(1 - Q_5)(1 - Q_6)(1 - Q_7) \\
& + (1 - Q_1)(1 - Q_2)(1 - Q_3)(1 - Q_4)(1 - Q_5)(1 - Q_6)(1 - Q_7)
\end{aligned} \tag{3}$$

The final formulation of the optimization problem for allocation reliability becomes as given in (4).

$$\text{Minimize } C(Q_1, \dots, Q_n) = \sum_{i=1}^n K_i \ln \left( \frac{1}{1 - (1 - Q_i)} \right), K_i > 0, i = 1, 2, \dots, n \tag{4}$$

Before solving the problem of assigning reliability to each part of the system, an initial evaluation can be performed to evaluate the outcome of the solution. This is achieved by calculating the reliability importance of each of the seven components of the system for each component. The (5) gives the formula:

$$I_Q(i) = \frac{\partial(1 - Q_i)}{\partial Q_i} \tag{5}$$

where this is used to calculate the reliability importance of each component, as shown in (6).

$$\begin{aligned}
\frac{\partial R_s}{\partial R_1} &= R_2 R_5 R_7 + R_3 R_5 R_7 + R_3 R_6 R_7 + R_4 R_6 R_7 - R_2 R_3 R_5 R_7 - R_3 R_4 R_6 R_7 \\
&\quad - R_3 R_5 R_6 R_7 - R_2 R_4 R_5 R_6 R_7 + R_2 R_3 R_4 R_5 R_6 R_7 \\
\frac{\partial R_s}{\partial R_2} &= R_1 R_5 R_7 - R_1 R_3 R_5 R_7 - R_1 R_4 R_5 R_6 R_7 + R_1 R_3 R_4 R_5 R_6 R_7 \\
\frac{\partial R_s}{\partial R_3} &= R_1 R_5 R_7 + R_1 R_6 R_7 - R_1 R_2 R_5 R_7 - R_1 R_4 R_6 R_7 - R_1 R_5 R_6 R_7 + R_1 R_2 R_4 R_5 R_6 R_7 \\
\frac{\partial R_s}{\partial R_4} &= R_1 R_6 R_7 - R_1 R_3 R_6 R_7 - R_1 R_2 R_5 R_6 R_7 + R_1 R_2 R_3 R_5 R_6 R_7 \\
\frac{\partial R_s}{\partial R_5} &= R_1 R_2 R_7 + R_1 R_3 R_7 - R_1 R_2 R_3 R_7 - R_1 R_3 R_6 R_7 - R_1 R_2 R_4 R_6 R_7 + R_1 R_2 R_3 R_4 R_6 R_7 \\
\frac{\partial R_s}{\partial R_6} &= R_1 R_3 R_7 + R_1 R_4 R_7 - R_1 R_3 R_4 R_7 - R_1 R_3 R_5 R_7 - R_1 R_2 R_4 R_5 R_7 + R_1 R_2 R_3 R_4 R_5 R_7 \\
\frac{\partial R_s}{\partial R_7} &= R_1 R_2 R_5 + R_1 R_3 R_5 + R_1 R_3 R_6 + R_1 R_4 R_6 - R_1 R_2 R_3 R_5 - R_1 R_3 R_4 R_6 \\
&\quad - R_1 R_3 R_5 R_6 - R_1 R_2 R_4 R_5 R_6 + R_1 R_2 R_3 R_4 R_5 R_6
\end{aligned} \tag{6}$$

Figure 4 depict the result of reliability allocation and importance for the ROSS of a spacecraft. Component 1 and Component 7 were among the most important components in the system used for this. Therefore, greater reliability and costs were allocated to these two components. As the failure of the component 1 or component 7 may lead to a complete failure of the system. By calculating the importance of reliability, the results are shown, in Table 1. Component 5 and component 6, as in the results obtained using the Jaya algorithm which was added to them with approximately the same reliability, based on the location of the component in the system. The value of the two composite reliability is also very close by using the importance of reliability equation. The reliability of component 3 was calculated, and the Jaya algorithm allocated a reliability of component 3 less than the two components 5 and the component 6, as shown in Table 1. Component 2, component 4, and as in the results obtained using Jaya algorithm, they have approximately the same reliability, depending on the location of the component in the system. The significance of the two-component reliability is also closely converged using the significance equation. The results shown in Table 1 that show the importance of each of the vehicles making up the system. The results can be arranged according to the importance of reliability and the position of the component in the system: (i) component 1, component 7, (ii) component 5, component 6, (iii) component 3. (iv) component 2, component 4. The results obtained can be illustrates using the Jaya algorithm shown in Table 1.

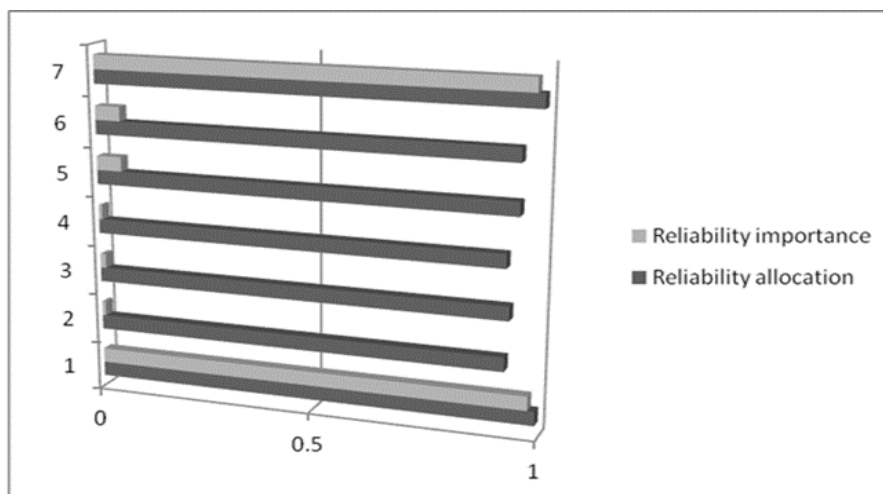


Figure 4. Reliability allocation and importance for the ROSS of a spacecraft

Table 1. Summary Table for optimal reliability allocation of ROSS of spacecraft components

Components	Reliability allocation	Reliability importance
Component 1	0.992	0.9762
Component 2	0.924	0.0086
Component 3	0.932	0.0129
Component 4	0.923	0.0084
Component 5	0.948	0.0582
Component 6	0.948	0.0582
Component 7	0.992	0.9762

#### 4. CONCLUSION

In this paper, the problem was formulated as a non-linear programming problem. The main objective of this paper is to reduce costs to the lowest possible level while ensuring increased reliability of the system (ROSS). The objective function is the cost function (the logarithmic function), which depends on the unreliability variables, as well as the constraints also depend on the unreliability variables. The problem was solved by using the Jaya algorithm, which proved its effectiveness in finding solutions in the fastest time compared to other algorithms and the accuracy of the solution.

#### ACKNOWLEDGEMENTS

The authors thank to AL-Iraqia University, University of Technology, Universiti Malaysia Pahang, Universiti Tun Hussein Onn and Universitas Ahmad Dahlan for supporting this collaborative research in the present work. This work was supported/funded by the Malaysia's Ministry of Higher Education under Fundamental Research Grant Scheme (FRGS/1/2018/ICT01/UTM/02/4).

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